



Lecture 11

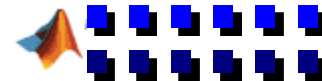
State Variable Feedback Design via Gain Transformation and Pole Placement

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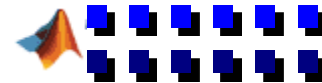
ECE 6095/4121
Modeling and Digital Control of Mechatronic Systems





SVFB Design via Gain Xformation & Pole Placement

1. **“Deadbeat” controller**
2. **Continuous-Discrete Gain Transformation**
 - Time response equivalence
 - Average gain method
 - Example-double integrator
3. **Pole Placement via SVFB Design**
 - Direct approach
 - Transformation approach
 - Ackermann formula/algorithm
 - Pole placement for MIMO systems
 - State feedback
 - output feedback (numerically not robust. So, won't discuss)
4. **Example-Inverted Pendulum on a Cart**
 - Continuous-discrete transformation design
 - Direct digital design
5. **Implementation of High-Order Compensators**
 - State prediction
 - Comparison with Smith compensator
 - Examples
6. **Command Inputs to SVFB Systems**
 - Integral feedback





Control in State-Space

- What can be done with respect to controlling system states?

$$\underline{x}(k+1) = \Phi \underline{x}(k) + \Gamma u(k)$$

$$y(k) = C\underline{x}(k)$$

$$\underline{x}(0) = \text{known initial condition}$$

Equivalent discrete system matrices

$$\Phi = e^{Ah}, \quad \Gamma = \int_0^h e^{A\sigma} d\sigma B$$

- State response

$$\underline{x}(k) = \Phi^k \underline{x}(0) + \sum_{i=0}^{k-1} \Phi^{k-1-i} \Gamma u(i)$$

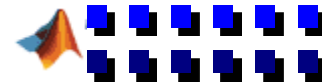
- Consider $k = n$

- Can we find $u(0), u(1), \dots, u(n-1)$ so that $\underline{x}(n) = \underline{\xi} =$ arbitrary vector, starting at any initial condition $\underline{x}(0)$?

$$\underline{\xi} - \Phi^n \underline{x}(0) = \sum_{i=0}^{n-1} \Phi^{n-1-i} \Gamma u(i) = \Gamma u(n-1) + \Phi \Gamma u(n-2) + \dots + \Phi^{n-1} \Gamma u(0)$$

$$= \underbrace{\begin{bmatrix} | & | & & | \\ \Gamma & \Phi\Gamma & \dots & \Phi^{n-1}\Gamma \\ | & | & & | \end{bmatrix}}_{H_c} \begin{bmatrix} u(n-1) \\ u(n-2) \\ \vdots \\ u(0) \end{bmatrix}$$

- If H_c is invertible, it is possible to find the requisite $\{u(i)\}$.
- Note: state may not necessarily stay at $\underline{\xi}$ for $k > n$.





Deadbeat Controller

- If system is cc the sequence $\{u(0), u(1), \dots, u(n-1)\}$ will drive $\underline{x}(0) \rightarrow \underline{x}(n) = \underline{\xi}$ where

$$\begin{bmatrix} u(n-1) \\ u(n-2) \\ \vdots \\ u(0) \end{bmatrix} = H_c^{-1} [\underline{\xi} - \Phi^n \underline{x}(0)]$$

- Open-loop control

$$\left. \begin{aligned} u(0) &= [0 \ 0 \ \dots \ 0 \ 1] H_c^{-1} [\underline{\xi} - \Phi^n \underline{x}(0)] \\ u(1) &= [0 \ 0 \ \dots \ 1 \ 0] H_c^{-1} [\underline{\xi} - \Phi^n \underline{x}(0)] \\ &\vdots \\ u(n-1) &= [1 \ 0 \ 0 \ \dots \ 0] H_c^{-1} [\underline{\xi} - \Phi^n \underline{x}(0)] \end{aligned} \right\} \underline{x}(0), \text{ \underline{not} } \underline{x}(k) \text{ is used here}$$

- Closed-loop control via time-invariance

"turn system on" at time "k": $\underline{x}(0) \Leftrightarrow \underline{x}(k), u(0) \Leftrightarrow u(k)$

$$\Rightarrow u(k) = [0 \ 0 \ \dots \ 1] H_c^{-1} [\underline{\xi} - \Phi^n \underline{x}(k)]$$

accomplishes same control sequence but via SVFB

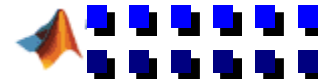
- Special case $\underline{\xi} = \underline{0}$

$$u(k) = - \overbrace{[0 \ 0 \ \dots \ 1]}^K H_c^{-1} \Phi^n \underline{x}(k)$$

is an SVFB control that reduces any (initial) state to $\underline{0}$ in n steps \rightarrow "deadbeat controller" (unique to discrete systems)

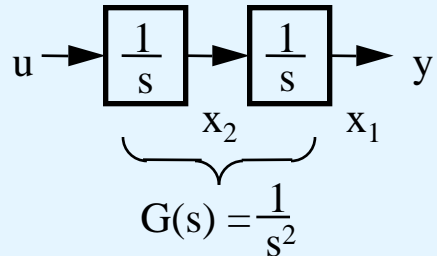
- CL dynamics $\underline{x}(k+1) = (\Phi - \Gamma K) \underline{x}(k)$
 $\underline{x}(n) = (\Phi - \Gamma K)^n \underline{x}(0) = \underline{0}$

$\Rightarrow \Phi - \Gamma K$ has all eigenvalues at $z = 0$





Example - Pure Inertia Control (e.g. Satellite)



$$\begin{bmatrix} \dot{x}_1(t) \\ \dot{x}_2(t) \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u(t)$$

$$y(t) = [1 \quad 0] \underline{x}(t)$$

- Equivalent discrete system

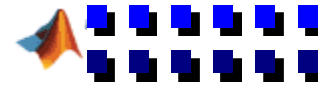
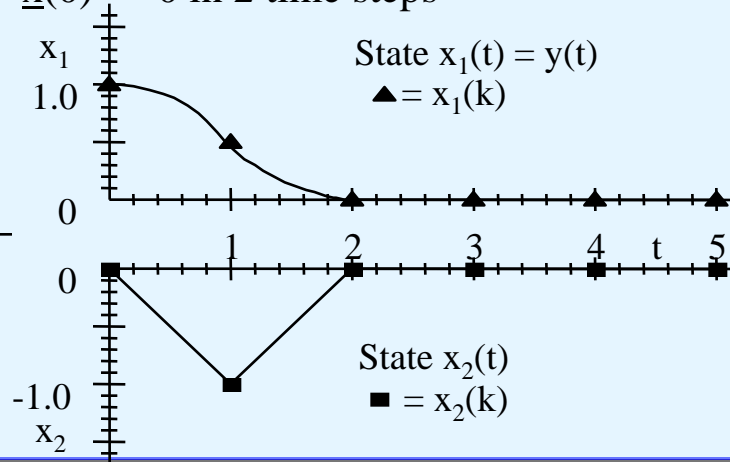
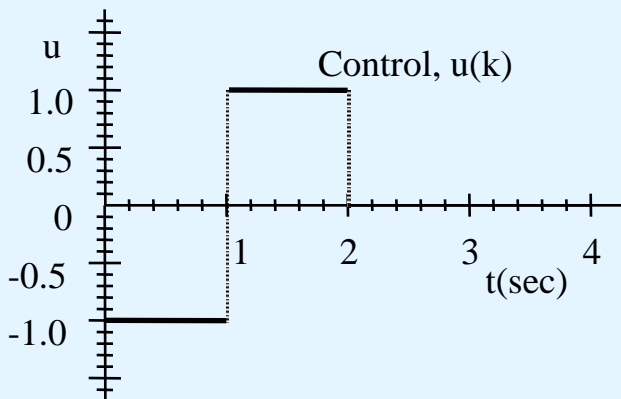
$$\underline{x}(k+1) = \underbrace{\begin{bmatrix} 1 & h \\ 0 & 1 \end{bmatrix}}_{\Phi} \underline{x}(k) + \underbrace{\begin{bmatrix} h^2/2 \\ h \end{bmatrix}}_{\Gamma} u(k)$$

Pick $h = 1$

- Deadbeat controller, $u(k) = -K \underline{x}(k) = -K_1 x_1(k) - K_2 x_2(k)$

$$K = [0 \quad 1] \begin{bmatrix} 1/2 & 3/2 \\ 1 & 1 \end{bmatrix}^{-1} \begin{bmatrix} 1 & 2 \\ 0 & 1 \end{bmatrix} = [1 \quad 3/2]$$

- Time response with $x_1(0) = 1, x_2(0) = 0$: $\underline{x}(0) \rightarrow 0$ in 2 time steps





Discrete SVFB Design Methods

- Continuous \rightarrow discrete equivalence methods

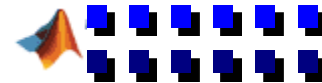
Given a continuous FB control law

$$u(t) = K_r r(t) - K \underline{x}(t)$$

develop from K_r, K an "equivalent" discrete control

$$u(k) = \tilde{K}_r r(k) - \tilde{K} \underline{x}(k)$$

- Idea: capitalize on earlier design for $\dot{\underline{x}} = A \underline{x} + B u$
- Compare continuous vs. discrete time response, phase margin, closed-loop poles, etc.
- Direct digital controller design
Find $u(k) = K_r r(k) - K \underline{x}(k)$ directly to place poles at $z = z_1, z_2, \dots, z_n$
 - (1) select $z_i = e^{s_i h}$ $i = 1, 2, \dots, n$;
 $s_i =$ desired pole location in s-plane
 - or, (2) select z_1, z_2, \dots, z_n directly
- Evaluation
 - Time response via simulation
 - Phase margin via Bode/Nyquist plot of
$$LG(j\omega) = K(zI - \Phi)^{-1} \Gamma \Big|_{z = e^{j\omega h}}$$
 - Sensitivity to parameters, RD, etc.





Continuous \rightarrow Discrete Gain Transformation Methods

$$C: \begin{cases} \dot{\underline{x}}(t) = A \underline{x}(t) + B u(t) \\ u(t) = K_r r(t) - K \underline{x}(t) \end{cases} \longrightarrow D: \begin{cases} \underline{x}(k+1) = \Phi \underline{x}(k) + \Gamma u(k) \\ u(k) = \underline{\tilde{K}}_r r(k) - \underline{\tilde{K}} \underline{x}(k) \end{cases}$$

Closed-loop

$$\dot{\underline{x}}(t) = (A - B K) \underline{x}(t) + K_r B r(t) \quad \underline{x}(k+1) = (\Phi - \Gamma \tilde{K}) \underline{x}(k) + \tilde{K}_r \Gamma r(k)$$

- Desire simplicity and accuracy
 - Time response of $D \stackrel{?}{\approx}$ time response of C
 - Eigenvalues of CL $D \stackrel{?}{\approx} \exp \{h \cdot \text{eigenvalues of CL } C\}$
- Start C at $\underline{x}(0)$ with $r(t) = r_0$. Response at $t = h$

$$\underline{x}(h) = e^{(A-BK)h} \underline{x}(0) + \int_0^h e^{(A-BK)\sigma} K_r B d\sigma \cdot r_0$$

- For discrete response

$$\underline{x}(h) = (\Phi - \Gamma \tilde{K}) \underline{x}(0) + \tilde{K}_r \Gamma r_0$$

A) Time response equivalence:

$$(1) \Phi - \Gamma \tilde{K} = e^{(A-BK)h} \rightarrow \text{"solve" for } \tilde{K} ?$$

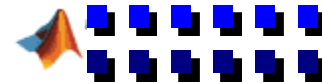
Only possible to obtain equivalence to $O(h^2)$

$$\tilde{K} \approx K + K(A-BK) h/2$$

$$(2) \tilde{K}_r \Gamma = K_r \int_0^h e^{(A-BK)\sigma} B d\sigma$$

Can only obtain equivalence to $O(h^2)$

$$\tilde{K}_r \approx \{1 - KB(h/2)\} K_r$$





B) Average Gain Method (Kleinman, Automatica, 1978)

- Consider C over $(0, h]$ with $\underline{x}(0)$, and $r(t) = r_0$

$$\underline{x}(t) = e^{(A-BK)t} \underline{x}(0) + \int_0^t e^{(A-BK)\sigma} K_r B d\sigma \cdot r_0$$

- Control, $u_c(t)$ over $(0, h]$, in continuous system

$$u_c(t) = K_r r_0 - K \underline{x}(t)$$

$$u_c(t) = \left[1 - K \int_0^t e^{(A-BK)\sigma} B d\sigma \right] K_r r_0 - K e^{(A-BK)t} \underline{x}(0)$$

- Discrete control over $(0, h] = u(0)$

$$u(0) = \tilde{K}_r r_0 - \tilde{K} \underline{x}(0)$$

$$\Rightarrow \text{Pick } \tilde{K}, \tilde{K}_r \text{ so that } u(0) = \bar{u}_c = \frac{1}{h} \int_0^h u_c(t) dt$$

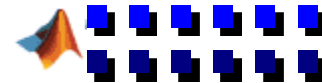
$$\bar{u}_c = \left[1 - \frac{K}{h} \int_0^h \int_0^t e^{(A-BK)\sigma} d\sigma dt B \right] K_r r_0 - \frac{K}{h} \int_0^h e^{(A-BK)t} dt \underline{x}(0)$$

$$\begin{aligned} \frac{K}{h} \int_0^h \int_0^t e^{(A-BK)\sigma} d\sigma dt &= \frac{K}{h} \int_0^h \bar{\Psi}(t) dt \\ &= \frac{K}{h} \left[\int_0^h (\bar{\Phi}(t) - I) dt \right] (A - BK)^{-1} \\ &= (\tilde{K} - K)(A - BK)^{-1} \end{aligned}$$

- Discrete equivalent gains

$$(1) \quad \tilde{K} = \frac{K}{h} \int_0^h e^{(A-BK)t} dt$$

$$(2) \quad \tilde{K}_r = \left[1 - \frac{K}{h} \int_0^h \int_0^t e^{(A-BK)\sigma} d\sigma dt B \right] K_r = \left[1 + (K - \tilde{K})(A - BK)^{-1} B \right] K_r$$





Computing Average Gain

- Obtain \tilde{K} using c2d, then compute \tilde{K}_r

- Approximation for small h

$$\tilde{K} \approx K \left[I + (A - BK) \frac{h}{2} + (A - BK)^2 \frac{h^2}{3!} + \dots \right] \quad \tilde{K}_r \approx \left\{ 1 - K \left[\frac{h}{2} + (A - BK) \frac{h^2}{3!} + \dots \right] B \right\} K_r$$

- Average gain scheme is "good" provided

$$h \leq \frac{1.0}{|\lambda_{\max}(A - BK)|} \sim \frac{1.0}{\|A - BK\|}$$

← Closed-loop system matrix

- Generally requires a smaller h than does the usual criterion

$$h \leq (0.5 \rightarrow 1.0) / |\lambda_{\max}(A)|$$

- Using average gain \tilde{K} is always better than just using K
 - but \tilde{K}_r may not maintain same DC gain as in continuous case
- Inverse procedure: given a discrete K_d , find continuous gain K.

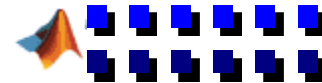
Solve

$$K = K_d h \left[\int_0^h e^{(A-BK)t} dt \right]^{-1}$$

iteratively:

$$K_{i+1} = K_d h \left[\int_0^h e^{(A-BK_i)t} dt \right]^{-1} \text{ with } K_0 = K_d.$$

- Generally converges in 2 - 3 iterations.
- Useful when h is subject to change, e.g., $K_d(h_1) \rightarrow K \rightarrow K_d(h_2)$





Example – Satellite Control, $G(s)=1/s^2$

$$\begin{bmatrix} \dot{x}_1(t) \\ \dot{x}_2(t) \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u(t); \quad y(t) = [1 \quad 0] \underline{x}(t)$$

Continuous SVFB control $\begin{matrix} \uparrow \\ K_r \end{matrix}$ $u(t) = 1.0 r(t) - \underbrace{[1 \quad 1]}_K \underline{x}(t)$

Gives CL system

$$\dot{\underline{x}}(t) = \begin{bmatrix} 0 & 1 \\ -1 & -1 \end{bmatrix} \underline{x}(t) + \begin{bmatrix} 0 \\ 1 \end{bmatrix} r(t); \quad y(t) = [1 \quad 0] \underline{x}(t)$$

CL transfer function

$$\frac{y(s)}{r(s)} = \frac{1}{s^2 + s + 1} \quad \left\{ \begin{array}{l} \text{CL poles at } s = -\frac{1}{2} \pm j\frac{\sqrt{3}}{2} \\ (\zeta = 0.5) \end{array} \right.$$

- Equivalent discrete gains, \tilde{K} , with $h = 0.5$

OK on "safety" requirement $h \leq \frac{1.0}{|\lambda_{\max}(A - BK)|} = \frac{1.0}{\sqrt{1}} = 1.0$

$$\tilde{K} = \frac{K}{h} \int_0^h e^{(A-BK)t} dt = [0.755 \quad 0.964]; \quad \tilde{K}_r = [1 + (K - \tilde{K})(A - BK)^{-1} B] K_r = 0.755$$

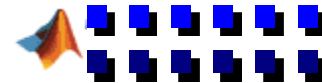
- Examine CL discrete eigenvalues (poles) of $\Phi - \Gamma\tilde{K}$

- "Expect" poles at $z_i = e^{s_i h} = e^{-0.25 \pm j0.433} = 0.707 \pm j0.327$

$$\Phi - \Gamma\tilde{K} = \begin{bmatrix} 1 & 0.5 \\ 0 & 1 \end{bmatrix} - \begin{bmatrix} 0.125 \\ 0.5 \end{bmatrix} [0.755 \quad 0.964] = \begin{bmatrix} 0.906 & 0.379 \\ -0.378 & 0.518 \end{bmatrix}$$

eigenvalues at $\lambda_i = 0.712 \pm j0.325 \quad (\zeta \approx 0.5)$

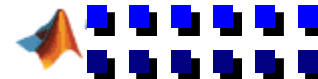
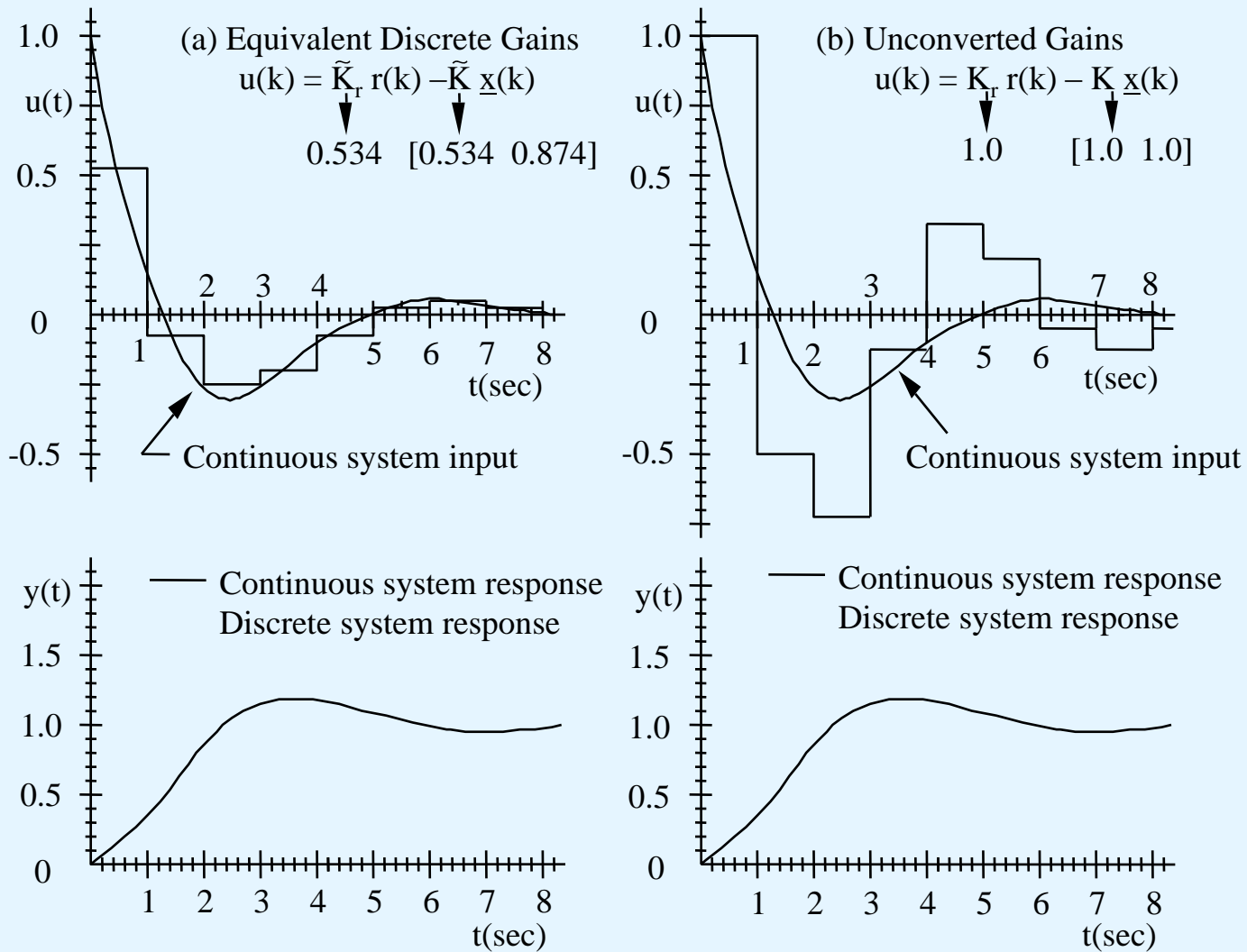
- Without gain equivalence $\lambda_i(\Phi - \Gamma K) = 0.688 \pm j0.39 \quad (\zeta = 0.41)$





Satellite Control Closed-Loop Simulation

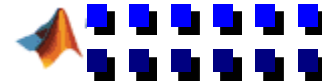
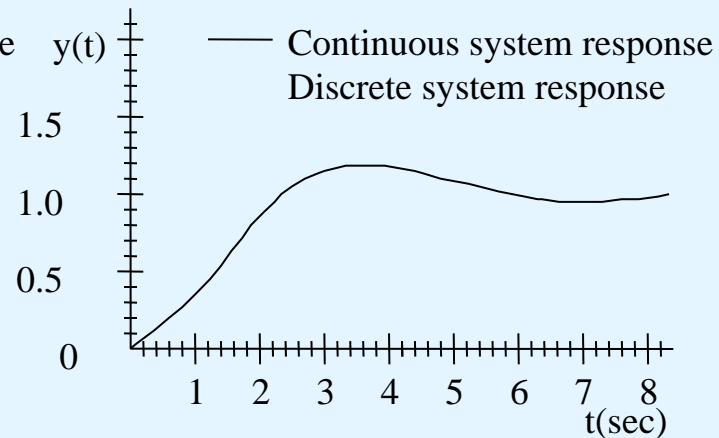
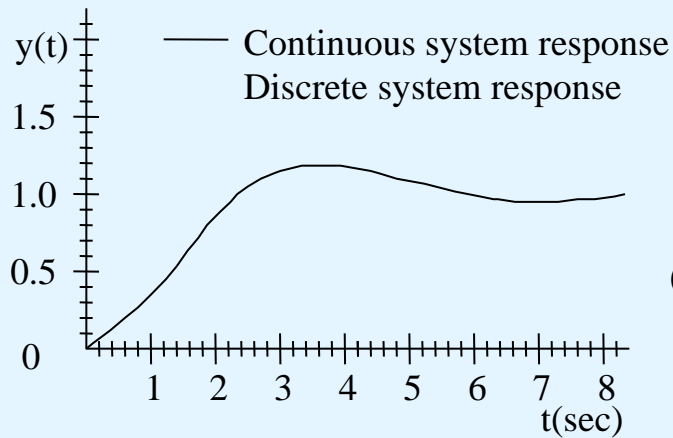
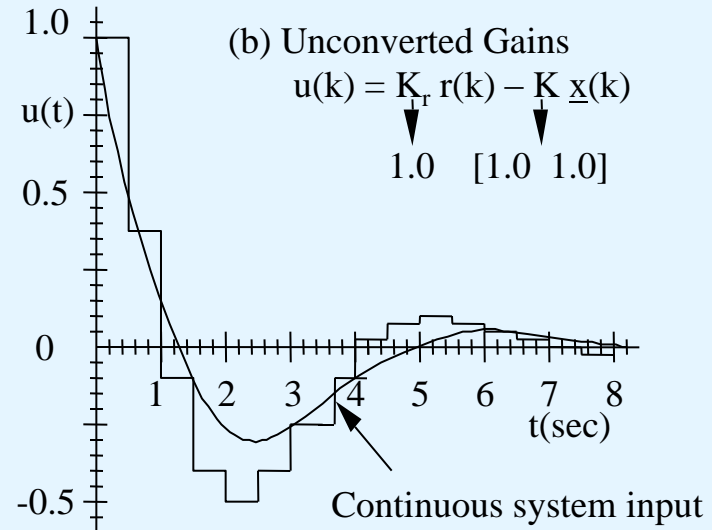
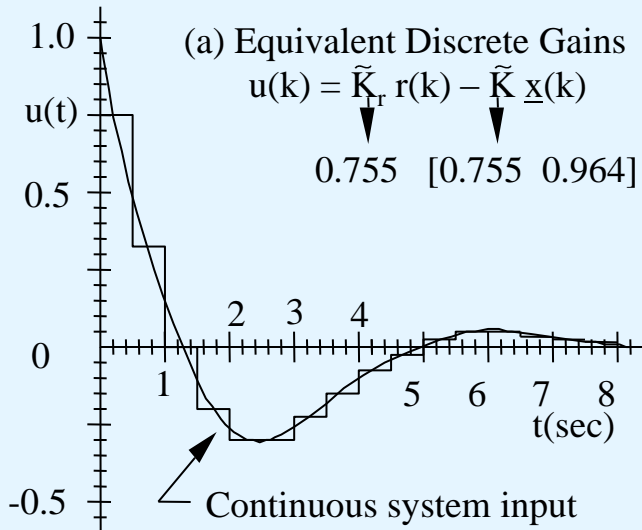
$\underline{x}(0)=\underline{0}$, $r(t)=1$, $h=1.0$, Average Gain Equivalence





Satellite Control Closed-Loop Simulation

$\underline{x}(0)=\underline{0}$, $r(t)=1$, $h=0.5$, Average Gain Equivalence

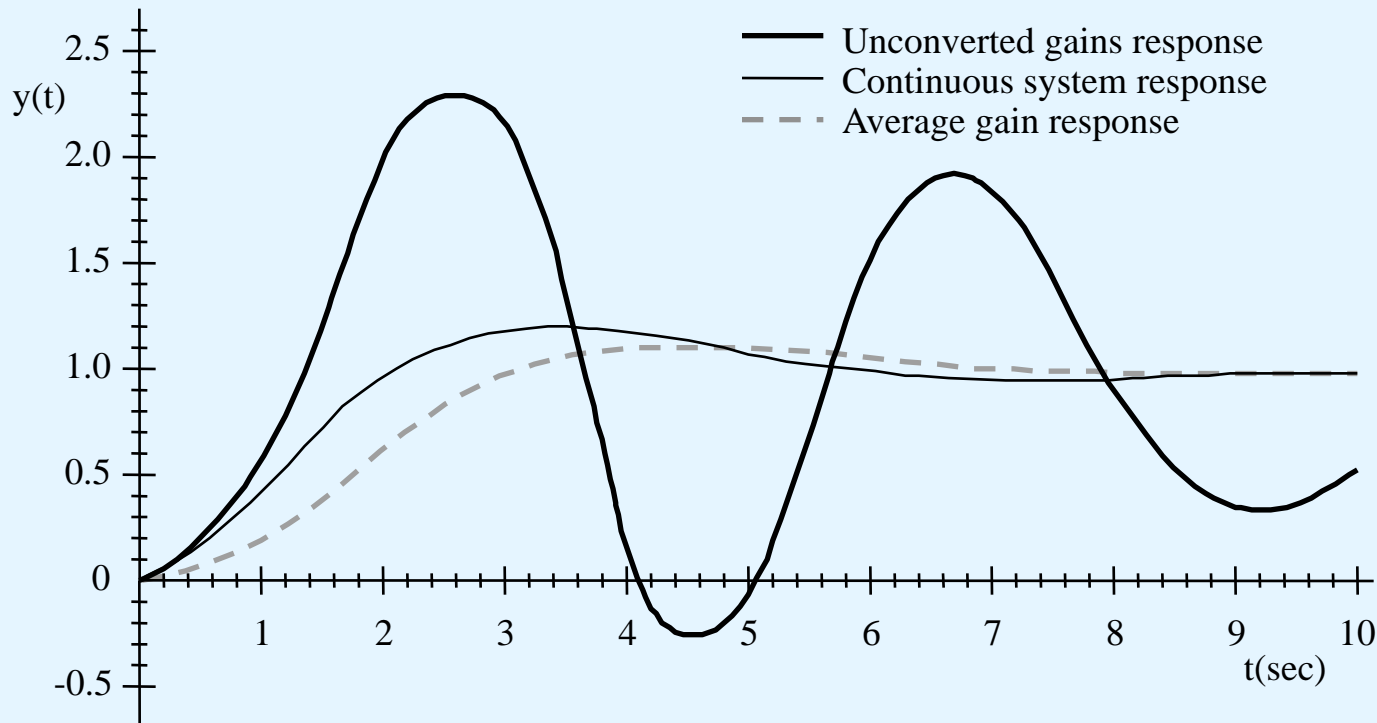




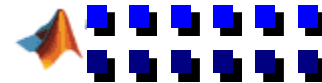
Satellite Control CL Simulation

$$\underline{x}(0)=[0 \ 0]', \ r(t)=1, \ h=1.8$$

- Even when $h > \frac{1.0}{|\lambda_{\max}(A - BK)|}$ the equivalent (average) gains will often give a "reasonable" CL system.
- With $h = 1.8$, $\tilde{K} = [0.261 \ 0.683]$, $\tilde{K}_r = 0.261$



- Unconverted discrete system ($K = [1 \ 1]$) becomes unstable as h increase.
- CL system with average gains still hanging in, with noticeable slow-down in step response.



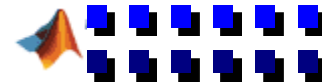


Summary of Equivalent Gain Method

- Average gain is best method to convert

$$K \rightarrow \tilde{K} ; \quad K_r \rightarrow \tilde{K}_r$$

- If $h \sim$ small ($< \frac{1.0}{|\lambda_{\max}(A - BK)|}$) use \tilde{K}, \tilde{K}_r
 - Do not simply use $\tilde{K} = K, \tilde{K}_r = K_r$ (instability as h increases).
 - Useful if need to change h on-line frequently
 - Store K, K_r from continuous design
 - Use series approximation to obtain \tilde{K}, \tilde{K}_r for current value of h
 - Generally
 - \tilde{K}_i will be smaller in magnitude than K_i .
 - Gains \tilde{K} will yield discrete CL poles with a slightly smaller ω_n than original continuous system (i.e., slower response).
 - Eigenvalues of $\Phi - \Gamma\tilde{K} \approx \exp(h \cdot \text{eigenvalues of } A - BK)$.
 - Phase margin of discrete system with average gain \approx phase margin of a discrete system with poles placed at $\exp[h \cdot \lambda_i(A - BK)]$.
 - DC gain ($r \rightarrow y$) of equivalent system not always same as C .
- \implies may wish to pick \tilde{K}_r so that DC gain of discrete CL system = DC gain of original continuous CL design, i.e., so that
- $$\tilde{K}_r C [I - \Phi + \Gamma \tilde{K}]^{-1} \Gamma = -K_r C [A - BK]^{-1} B$$
- If $h \neq$ small, design K, K_r directly for discrete system.





State Variable Feedback Control – Direct Pole Placement (SISO)

- Discrete system design

- Given

$$\underline{x}(k+1) = \Phi \underline{x}(k) + \Gamma u(k)$$

$$y(k) = C \underline{x}(k)$$

with $|zI - \Phi| = p(z) = z^n + a_1 z^{n-1} + \dots + a_n$ and linear feedback control structure

$$u(k) = K_r r(k) - K \underline{x}(k)$$

- Closed-loop dynamics

$$\underline{x}(k+1) = \underbrace{(\Phi - \Gamma K)}_{\bar{\Phi}} \underline{x}(k) + K_r \Gamma r(k)$$

$\bar{\Phi}$ = discrete closed-loop matrix

$$y(k) = C \underline{x}(k)$$

- Pole placement

(1) Find FB gains $K = [K_1, K_2, \dots, K_n]$ so that the closed-loop system matrix $\bar{\Phi}$ has eigenvalues (poles) at pre-selected locations z_1, z_2, \dots, z_n^*

$$\Rightarrow |zI - (\Phi - \Gamma K)| = (z - z_1) \cdots (z - z_n) = p_d(z)$$

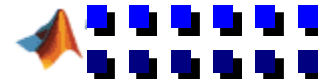
$$p_d(z) = z^n + d_1 z^{n-1} + \dots + d_n$$

↙ desired CL characteristic polynomial

2. Adjust K_r to have unity (or some desired) DC gain from r to y

$$\frac{y(z)}{r(z)} = K_r \left[C(zI - \bar{\Phi})^{-1} \Gamma \right]_{z=1} = 1$$

*Usually $z_i = e^{s_i h}$ where the $\{s_i\}$ are desired pole locations in s-plane.



SISO Pole Placement Methods

- Same scheme should work for either continuous or discrete problems, $\Phi \Leftrightarrow A$, $\Gamma \Leftrightarrow B$
 $\Phi - \Gamma K \Leftrightarrow A - BK$

- Direct approach

- Expand $|zI - \Phi + \Gamma K| = z^n + f_1(K)z^{n-1} + \dots + f_n(K)$

[each f_i will be linear in K_1, K_2, \dots, K_n]

- Expand $p_d(z) = (z - z_1)(z - z_2) \dots (z - z_n) = z^n + d_1z^{n-1} + \dots + d_n$

- Equate coefficients and solve n linear equations, n unknowns

$$f_i(K) = d_i; \quad i = 1, 2, \dots, n$$

- Useful in simple problems, some structured ones

- Example: $z_1 = 0.5 + j0.3$, $z_2 = 0.5 - j0.3$

$$\Phi = \begin{bmatrix} 1.0 & 0.2 \\ 0.2 & 1.0 \end{bmatrix} \quad \Gamma = \begin{bmatrix} 1.0 \\ 0.5 \end{bmatrix} \quad K = \begin{bmatrix} K_1 & K_2 \end{bmatrix}$$

$$\bar{\Phi} = \Phi - \Gamma K = \begin{bmatrix} 1.0 - K_1 & 0.2 - K_2 \\ 0.2 - 0.5K_1 & 1.0 - 0.5K_2 \end{bmatrix} \quad p_d = (z - z_1)(z - z_2) = z^2 - 1.0z + 0.34$$

$$|zI - \bar{\Phi}| = z^2 + \underbrace{(-2 + K_1 + 0.5K_2)}_{f_1} z + \underbrace{(0.96 - 0.3K_2 - 0.9K_1)}_{f_2}$$

$$\left. \begin{array}{l} -2 + K_1 + 0.5K_2 = -1 \\ 0.96 - 0.9K_1 - 0.3K_2 = 0.34 \end{array} \right\} \Rightarrow K_1 = 0.067, \quad K_2 = 1.867$$

- Select K_r (e.g. so that DC gain = 1)



Transformation Approach for Pole Placement

- Let $\underline{v}(k) = T^{-1}\underline{x}(k)$ where T transforms Φ, Γ to SCF

$$\underline{x}(k+1) = \Phi \underline{x}(k) - \Gamma K \underline{x}(k) \Rightarrow \underline{v}(k+1) = \underbrace{T^{-1}\Phi T}_{\text{SCF}} \underline{v}(k) - \underbrace{T^{-1}\Gamma K T}_{\text{SCF}} \underline{v}(k)$$

$$\begin{bmatrix} -a_1 & -a_2 & \dots & -a_n \\ 1 & 0 & \dots & 0 \\ 0 & 1 & \dots & \vdots \\ \vdots & \vdots & \ddots & \vdots \\ 0 & \dots & 1 & 0 \end{bmatrix} \begin{bmatrix} 1 \\ 0 \\ \vdots \\ \vdots \\ 0 \end{bmatrix}$$

a_i = coefficients of open-loop characteristic polynomial

- If $KT = [-a_1 + d_1, -a_2 + d_2, \dots, -a_n + d_n]$,

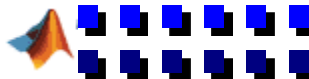
$$\text{then } T^{-1}\Phi T - T^{-1}\Gamma K T = T^{-1}(\Phi - \Gamma K)T = \begin{bmatrix} -d_1 & -d_2 & \dots & -d_n \\ 1 & 0 & \dots & 0 \\ 0 & 1 & \dots & \vdots \\ \vdots & \vdots & \ddots & \vdots \\ 0 & \dots & 1 & 0 \end{bmatrix}$$

$\Rightarrow \Phi - \Gamma K$ has desired characteristic polynomial $p_d(z)$ with

$$K = [-a_1 + d_1, -a_2 + d_2, \dots, -a_n + d_n] T^{-1}$$

- Best to solve

$$T^{-1} \begin{bmatrix} K_1 \\ K_2 \\ \vdots \\ K_n \end{bmatrix} = \begin{bmatrix} -a_1 + d_1 \\ -a_2 + d_2 \\ \vdots \\ -a_n + d_n \end{bmatrix}$$





Algorithm for Obtaining T

- Useful in general, not just for pole-placement problems

$$T = \begin{bmatrix} | & | & \cdots & | \\ \underline{t}_1 & \underline{t}_2 & \cdots & \underline{t}_n \\ | & | & \cdots & | \end{bmatrix}$$

- Generate T columnwise

$$\begin{aligned} \underline{t}_1 &= \Gamma \\ \underline{t}_2 &= \Phi \underline{t}_1 + a_1 \Gamma \\ \underline{t}_3 &= \Phi \underline{t}_2 + a_2 \Gamma \\ &\vdots \\ \underline{t}_n &= \Phi \underline{t}_{n-1} + a_{n-1} \Gamma \quad (\text{check: does } \Phi \underline{t}_n = -a_n \Gamma ?) \end{aligned}$$

- Requires computation of $\{a_i\}$ -- possible numerical problems
- T^{-1} will exist if system is completely controllable

column \underline{t}_k is a linear combination of $\Gamma, \Phi\Gamma, \dots, \Phi^{k-1}\Gamma$

- If $y(k) = C \underline{x}(k)$

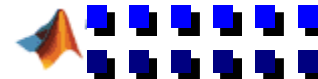
$$y(k) = CT \underline{y}(k) = [b_1, b_2, \dots, b_n] \underline{y}(k) \quad \text{with} \quad b_i = C \underline{t}_i$$

- Closed-loop transfer function with $u(k) = K_r r(k) - K \underline{x}(k)$

$$\frac{y(z)}{r(z)} = K_r \frac{b_1 z^{n-1} + \dots + b_n}{z^n + d_1 z^{n-1} + \dots + d_n} ; \quad b_i = C \underline{t}_i$$

- for unity DC gain = $K_r = \left(1 + \sum_{i=1}^n d_i \right) / \sum_{i=1}^n b_i$

- Loop gain: $K(zI - \Phi)^{-1} \Gamma = \frac{\gamma_1 z^{n-1} + \dots + \gamma_n}{z^n + a_1 z^{n-1} + \dots + a_n} ; \quad \gamma_i = K \underline{t}_i = d_i - a_i$



Ackermann Formula

- Circumvents requirement to compute a_i

$p_d(z) = (z - z_1) (z - z_2) \cdots (z - z_n) = z^n + d_1 z^{n-1} + \cdots + d_n$ desired CL characteristic polynomial

$$K = [0 \ 0 \ \dots \ 1] H_c^{-1} p_d(\Phi)$$

$$H_c = \begin{bmatrix} | & | & \dots & | \\ \Gamma & \Phi\Gamma & \dots & \Phi^{n-1}\Gamma \\ | & | & \dots & | \end{bmatrix} = \text{Controllability matrix}$$

$$p_d(\Phi) = (\Phi - z_1 I) (\Phi - z_2 I) \cdots (\Phi - z_n I)$$

(z_i = desired poles, must be in complex conjugate pairs)

- Algorithm

1. Set up H_c matrix one column at a time. Transpose $H_c \rightarrow H_c'$.

Numerically unstable

2. Solve

$$H_c' \mathbf{q} = \begin{bmatrix} 0 \\ 0 \\ \vdots \\ 1 \end{bmatrix} \text{ for n-vector } \mathbf{q}$$

- Use any available routine for solving $A \underline{x} = \underline{b}$.

- If solution fails, stop. System is not completely controllable.

3. Evaluate $p_d(\Phi) = X$.

4. Obtain gains $K = \mathbf{q}' X = [K_1, K_2, \dots, K_n]$

5. Compute K_r if needed.



Algorithm to Obtain $P_d(\Phi)$

- Compute using complex conjugate pairs to avoid complex matrices

- e.g., if $z_3 = a + bj$, $z_4 = a - bj$

$$(z - z_3)(z - z_4) = z^2 - 2az + (a^2 + b^2)$$

$$(\Phi - z_3 I)(\Phi - z_4 I) = \Phi^2 - 2a\Phi + (a^2 + b^2)I$$

- Incorporate into iterative scheme

1. initialize $X = I$, $k = 1$
2. read real & imag. parts of roots $RA(i)$, $RB(i)$, $i = 1, 2, \dots, n$
3. if $RB(k) = 0$: $X = X * [\Phi - RA(k)I]$
 $k = k + 1$
4. if $RB(k) \neq 0$: $X = X * [\Phi^2 - 2RA(k)\Phi + (RA^2(k) + RB^2(k))I]$
 $k = k + 2$
5. if $k \leq n$ go to 3 ; if $k = n + 1$ done

- Develop as a Subroutine GAINS
 - Can be used for continuous or discrete models

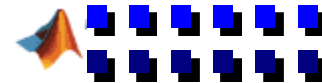
- Generally pick z_i via $e^{s_i h}$

- No restriction on h other than usual $\frac{(0.5 \rightarrow 1.0)}{|\lambda_{\max}(A)|}$

- Deadbeat response: all $z_i = 0 \Rightarrow p_d(z) = z^n$; also $p_d(\Phi) = \Phi^n$

$$K = [0 \ 0 \ \dots \ 1] H_c^{-1} \Phi^n$$

- deadbeat gains $K_i \rightarrow \infty$ as $h \rightarrow 0$





Example – Satellite Control/Pointing

$$\underline{x}(k+1) = \begin{bmatrix} 1 & h \\ 0 & 1 \end{bmatrix} \underline{x}(k) + \begin{bmatrix} h^2/2 \\ h \end{bmatrix} u(k)$$

- Desired CL characteristic polynomial:

$$z^2 + d_1 z + d_2 = p_d(z)$$

- if $s_i = -\zeta\omega_n \pm j\omega_n\sqrt{1-\zeta^2}$: $d_1 = -2e^{-\zeta\omega_n h} \cos(\omega_n h\sqrt{1-\zeta^2})$; $d_2 = e^{-2\zeta\omega_n h}$

- Use Ackermann algorithm: $K = [0 \ 1] H_c^{-1} p_d(\Phi)$

$$H_c = \begin{bmatrix} | & | \\ \Gamma & \Phi\Gamma \\ | & | \end{bmatrix} = \begin{bmatrix} h^2/2 & 3h^2/2 \\ h & h \end{bmatrix}; \text{ solve } \underbrace{\begin{bmatrix} h^2/2 & h \\ 3h^2/2 & h \end{bmatrix}}_{H_c'} \begin{bmatrix} q_1 \\ q_2 \end{bmatrix} = \begin{bmatrix} 0 \\ 1 \end{bmatrix} \Rightarrow \underline{q} = \begin{bmatrix} 1/h^2 \\ -0.5/h \end{bmatrix}$$

$$p_d(\Phi) = \begin{bmatrix} 1+d_1+d_2 & 2h+d_1h \\ 0 & 1+d_1+d_2 \end{bmatrix} = \Phi^2 + d_1\Phi + d_2I$$

$$K = \underline{q}' p_d(\Phi) = \begin{bmatrix} \frac{1+d_1+d_2}{h^2} & \frac{3+d_1-d_2}{2h} \end{bmatrix}$$

- Deadbeat controller: $d_1 = d_2 = 0 \implies K = [1/h^2 \ 3/2h]$

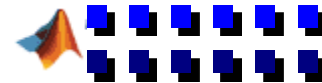
- as $h \rightarrow 0$, require excessive control energy

- good control scheme when h is large

- For $\zeta = 0.707$, $\omega_n = 0.707$, $h = 0.5$

$$s_i = -0.5 \pm j0.5 \rightarrow z_i = e^{s_i h} = 0.53 \pm j0.29$$

$$d_1 = -1.5092, \ d_2 = 0.6065 \rightarrow K = [0.3894 \ 0.8843]$$



RQ Implementation

- Transform (Φ, Γ) to controller Hessenberg form, $H_p = Q^T \Phi Q$; $\Gamma_p = Q^T \Gamma = \beta \underline{e}_1$
- Perform RQ factorization of Hessenberg as follows

Set $H_1 = H_p$

For $i=1,2,\dots,n$ Do

$$R_i Q_i = H_i - \lambda_i I$$

$$H_{i+1} = Q_i R_i + \lambda_i I$$

End

Note $H_{i+1} = Q_i H_i Q_i^T \Rightarrow H_{n+1} = Q_n Q_{n-1} \dots Q_2 Q_1 H_p Q_1^T Q_2^T \dots Q_n^T Q_n^T$

$$\Rightarrow P_d(H_p) = \prod_{i=1}^n (H_p - \lambda_i I) = R_1 R_2 \dots R_n Q_n Q_{n-1} \dots Q_2 Q_1$$

$$\Rightarrow P_d(\Phi) = Q P_d(H_p) Q^T$$

Controllability matrix: $H_c = Q \begin{bmatrix} \Gamma_p & H_p \Gamma_p & \dots & H_p^{n-1} \Gamma_p \end{bmatrix}$

$$K = \underline{e}_n^T H_c^{-1} P_d(\Phi) = \underline{e}_n^T \underbrace{\begin{bmatrix} \Gamma_p & H_p \Gamma_p & \dots & H_p^{n-1} \Gamma_p \end{bmatrix}^{-1}}_{\text{upper } \Delta} R_1 R_2 \dots R_n Q_n Q_{n-1} \dots Q_2 Q_1 Q^T$$

$$= \underbrace{\alpha}_{\alpha} \prod_{i=1}^n [R_i]_{nn} \underline{e}_n^T Q_n Q_{n-1} \dots Q_2 Q_1 Q^T; \alpha' = 1 / \prod_{i=1}^{n-1} [H_p]_{i+1,i}$$

- Algorithm

1. Set $H_1 = H_p$, $\alpha = \left(\prod_{i=1}^{n-1} [H_p]_{i+1,i} \right)^{-1}$

2. For $i=1,2,\dots,n$ DO

$$R_i Q_i = H_i - \lambda_i I$$

$$H_{i+1} = Q_i R_i + \lambda_i I$$

$$\alpha \rightarrow \alpha [R_i]_{nn}$$

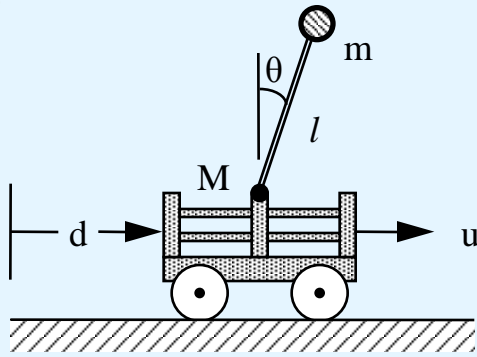
End

3. Obtain gains

$$K = \alpha \underline{e}_n^T Q_n Q_{n-1} \dots Q_2 Q_1 Q^T$$

Numerically stable
Implementation of 'place' command

Example – Inverted Pendulum on a Cart



Small angle equations:

$$\ddot{\theta}(t) = \frac{(m + M)g}{Ml} \theta(t) - \frac{u(t)}{Ml}$$

$$\ddot{d}(t) = -g \frac{m}{M} \theta(t) + \frac{u(t)}{M}$$

State equation $\underline{\dot{x}} = [\theta, q, d, v]'$; $q = \dot{\theta}$, $v = \dot{d}$

$$\underline{\dot{x}} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ \frac{(m+M)g}{Ml} & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \\ -g \frac{m}{M} & 0 & 0 & 0 \end{bmatrix} \begin{bmatrix} \theta \\ q \\ d \\ v \end{bmatrix} + \begin{bmatrix} 0 \\ -\frac{1}{Ml} \\ 0 \\ \frac{1}{M} \end{bmatrix} u$$

Let $m = 0.1$, $M = 1.0$, $l = 1\text{m}$, $g \approx 10\text{m/sec}^2$

$$\underline{\dot{x}} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ 11 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \\ -1 & 0 & 0 & 0 \end{bmatrix} \underline{x} + \begin{bmatrix} 0 \\ -1 \\ 0 \\ 1 \end{bmatrix} u$$

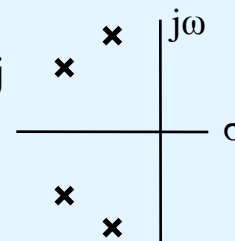
Open-loop eigenvalues at $s = 0, 0, \sqrt{11}, -\sqrt{11}$

- Objective - Find SVFB $u = -K\underline{x}$ such that any $\underline{x}(0) \rightarrow \underline{0}$ (regulator design, $r = 0$)
- Continuous time design: $u(t) = -[77.9 \ -23.0 \ -16.9 \ -13.0] \underline{x}(t)$ gives a stable CL system

$\underline{\dot{x}} = (A - BK) \underline{x}$ with CL eigenvalues

$$\lambda_1(A - BK) = -2 \pm 3j, \quad -3 \pm 2j$$

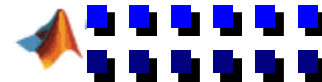
- Examine discrete time design(s)
 - equivalent (average) gains
 - direct design with $z_i = e^{s_i h}$



$$G(s)G(-s) = \frac{1}{1 + (-1)^n \left(\frac{s}{\omega_c}\right)^{2n}} = \frac{1}{1 + \left(\frac{-s^2}{\omega_c^2}\right)^n}$$

$$\text{Roots: } \left(\frac{-s^2}{\omega_c^2}\right) = (-1)^{1/n} \Rightarrow \left(\frac{-s^2}{\omega_c^2}\right) = e^{\frac{j(2k-1)\pi}{n}}$$

$$s_k = \omega_c e^{\frac{j(2k-1+n)\pi}{2n}}; k=1, 2, \dots, n$$



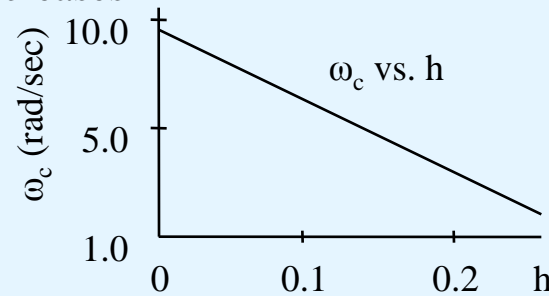
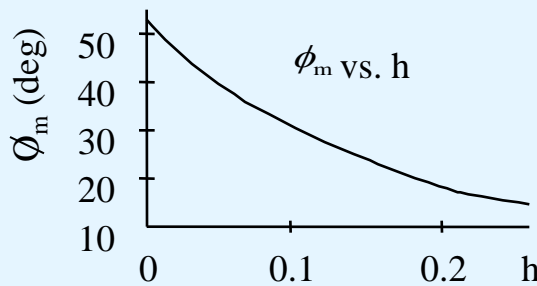


Equivalent Discrete Design, $u(k) = -\tilde{K}\underline{x}(k)$

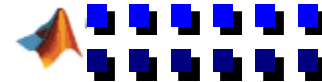
- Expect good performance for $h \leq \frac{1.0}{|\lambda_{\max}(A - BK)|} = \frac{1}{\sqrt{13}} = 0.28$
- System parameters vs. h (sec)

h	\tilde{K}_1	\tilde{K}_2	\tilde{K}_3	\tilde{K}_4	ω_c	ϕ_m
0.00	-77.9	-23.0	-16.9	-13.0	9.4	53.6°
0.02	-72.6	-21.5	-15.3	-11.9	9.1	46.7
0.05	-65.1	-19.4	-13.0	-10.3	8.6	39.8
0.10	-53.5	-16.0	-9.53	-7.92	7.4	30.0
0.15	-43.2	-13.0	-6.61	-5.81	6.5	23.1
0.20	-34.1	-10.4	-4.17	-3.98	5.4	17.7
0.25	-26.3	-8.05	-2.16	-2.42	4.6	14.8

- Large decrease in gain magnitudes as h increases



- Stability analysis
 - Discrete system has a delay of $\sim h/2$ sec \Rightarrow Reduces ϕ_m by $\omega_c h/2$
 - To avoid instability, average gain lowers ω_c
 - \Rightarrow Lessens destabilizing effect of discretization delay
 - \Rightarrow Moves ω_c into a region where $\angle LG(j\omega)$ is larger
 - Examine Bode plot of LG to get ω_c and ϕ_m $LG(j\omega) = \tilde{K}(zI - \Phi)^{-1} \Gamma|_{z=e^{j\omega h}}$

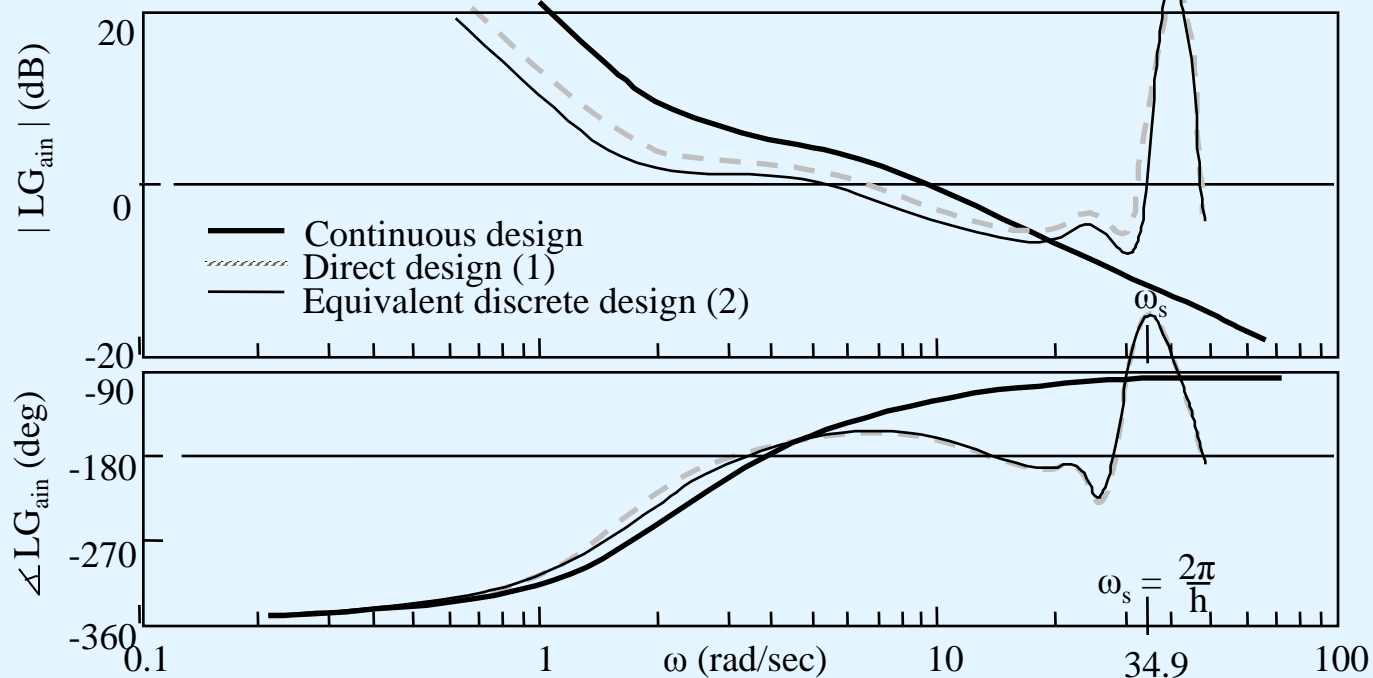


Direct Digital Design - Inverted Pendulum

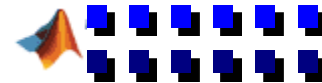
- Selection of $h \sim \frac{0.5 - 1}{|\lambda_{\max}(A)|} = \frac{.5 - 1}{\sqrt{11}} = 0.15$ to 0.3 sec

Pick $h = 0.18$ sec

- Pole placement @ $z_i = e^{s_i h} = \{ 0.60 \pm j0.36, 0.55 \pm j0.21 \}$
 \Rightarrow discrete gains $K = [-43.8 \quad -13.2 \quad -6.67 \quad -5.91]$
- Equivalent discrete design: $\tilde{K} = [-37.6 \quad -11.4 \quad -5.09 \quad -4.68]$
 yields closed-loop poles of $\Phi - \Gamma \tilde{K} = \{ 0.67 \pm j0.41, 0.57 \pm j0.14 \}$



- ω_c pole placement = 6.7 vs. 5.9 for equivalent design. Both have \sim same $\phi_m \approx 19.3^\circ$!
 Continuous system had 56.3° .



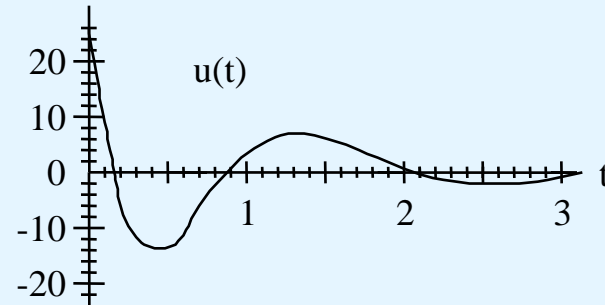
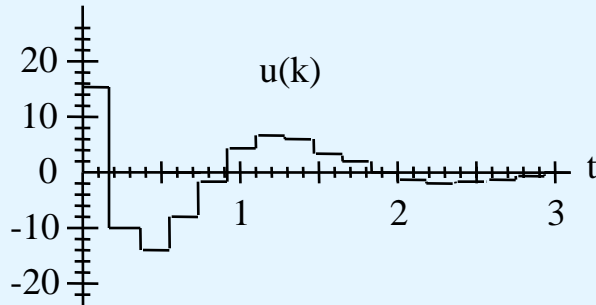


Closed-Loop Simulaton, Inverted Pendulum, $\underline{x}(0)=[0.2, 0, 1, 0]'$

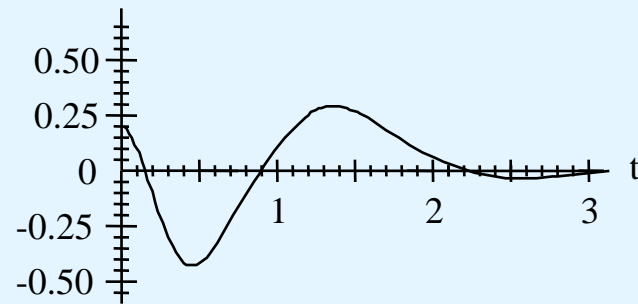
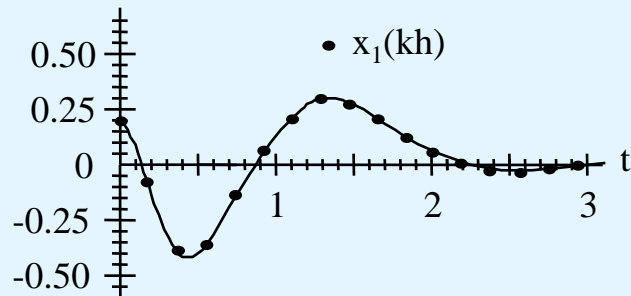
DIGITAL CONTROL #1 ($h = 0.18$)

CONTINUOUS CONTROL

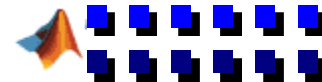
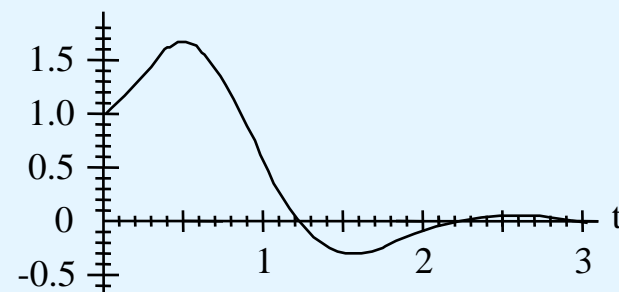
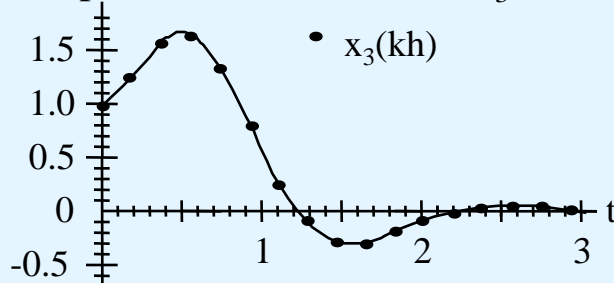
(a) Control input



(b) Pendulum angle from vertical, $x_1(t)$, rad

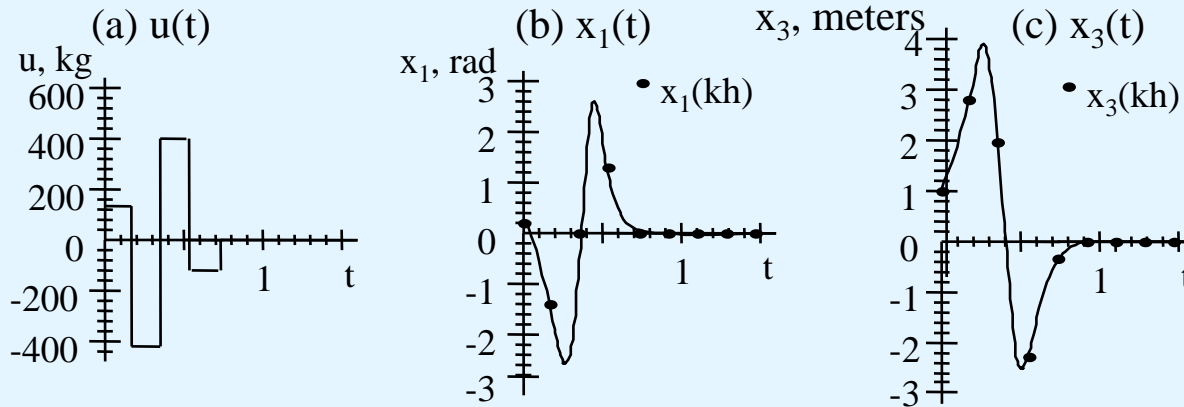


(c) Cart position from "home", $x_3(t)$, meters

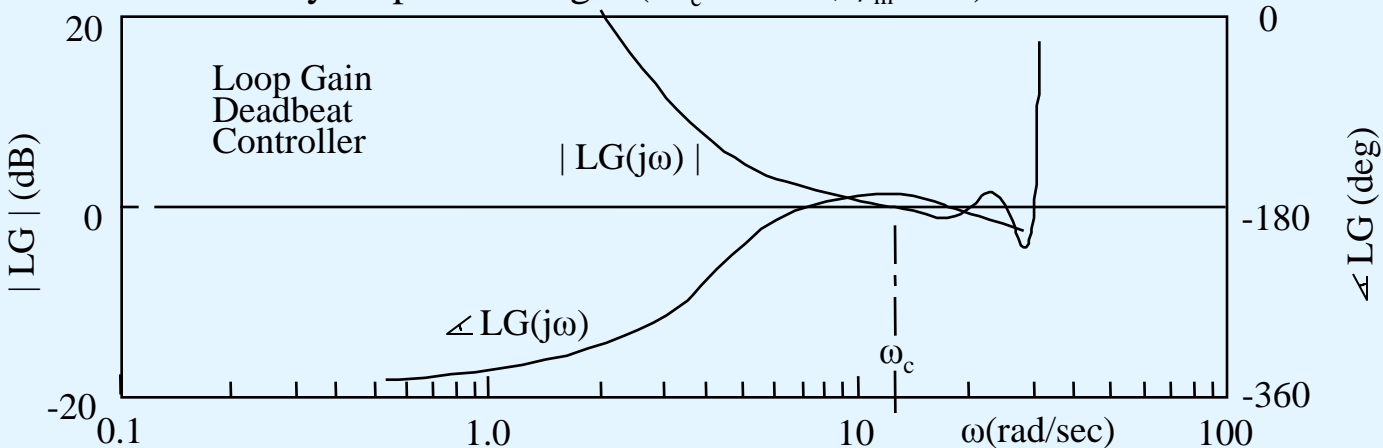


Deadbeat Controller, Inverted Pendulum, $\underline{x}(0)=[0.2, 0, 1, 0]'$

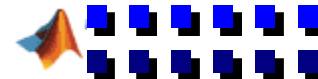
- Impossible physically in this example, but interesting... gains $K = [-190.92 \ -53.52 \ -92.48 \ -41.62]$



- Excessive control and state overshoots. Yet $\underline{x}(4) = \underline{0}$.
- And there is virtually no phase-margin ($\omega_c \approx 13.3$, $\phi_m \approx 4^\circ$).



- CL system has virtually no robustness properties.
- Best to reserve deadbeat for "slow" systems with $h \sim$ large.



SVFB with Time Delay in Control, $\tau = Mh + \epsilon$

- First design SVFB $u(k) = -K\underline{x}(k)$ assuming $\tau = 0$.

Case 1: $M > 0, \epsilon = 0$

$$\underline{x}(k+1) = \Phi \underline{x}(k) + \Gamma u(k-M)$$

- Predictor controller

$$u(k) = -K \hat{\underline{x}}(k+M) \quad \begin{array}{l} \text{prediction of state at time } (k + M)h \\ \text{from } \underline{x}(k) \text{ and } u(k-1), \dots, u(k-M) \end{array}$$

$$\hat{\underline{x}}(k+1) = \Phi \underline{x}(k) + \Gamma u(k-M)$$

$$\hat{\underline{x}}(k+2) = \Phi \hat{\underline{x}}(k+1) + \Gamma u(k-M+1)$$

$$\vdots = \Phi^2 \underline{x}(k) + \Phi \Gamma u(k-M) + \Gamma u(k-M+1)$$

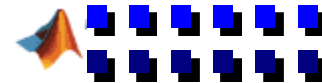
$$\hat{\underline{x}}(k+M) = \Phi^M \underline{x}(k) + \sum_{i=1}^M \Phi^{i-1} \Gamma u(k-i)$$

- Present control $u(k)$ will have its first effect on $\underline{x}(k+1+M)$
- Need to store past controls in a pushdown stack
- Requires a good knowledge of Φ, Γ to perform accurate propagation of $\underline{x}(k)$

Case 2: $M = 0, \epsilon > 0$ ($0 \leq \epsilon < h$)

- Use $u(k) = -K \hat{\underline{x}}(kh + \epsilon)$
 $\hat{\underline{x}}(kh + \epsilon) = e^{A\epsilon} \underline{x}(k) + \int_0^\epsilon e^{A\sigma} d\sigma B u(k-1)$
 $\Rightarrow u(k) = -K_x \underline{x}(k) - K_u u(k-1)$

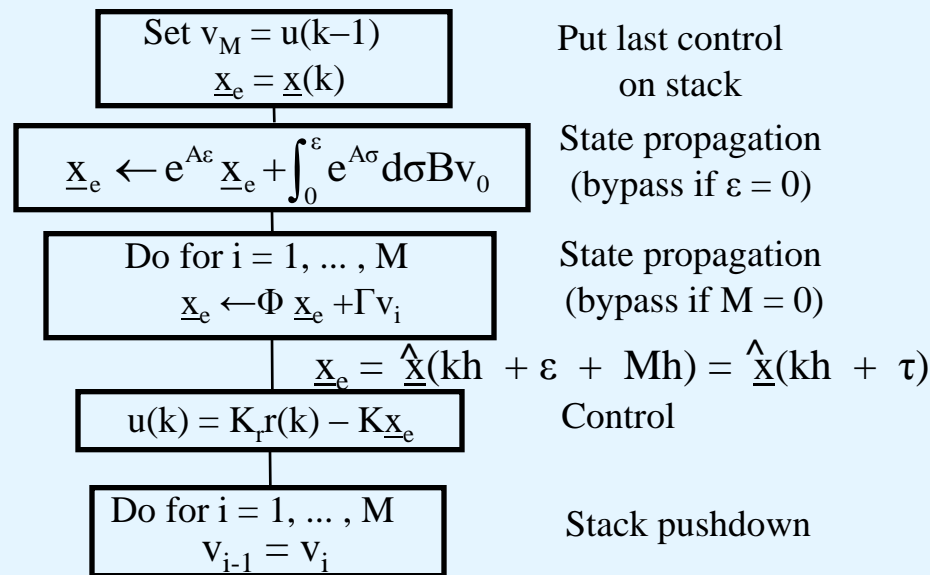
- Modification to structure only, propagation "hidden"
- Identical to earlier equations when $\epsilon = h^-$ (corresp to $M = 1$)



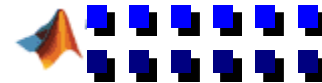
Implementation of Delay Compensator, General Case

- Basic Idea: construct $u(k)$ so that τ sec later, $u(kh + \tau) \approx -K\underline{x}(kh + \tau)$
 \Rightarrow input $u(k) = -K\hat{\underline{x}}(kh + \tau)$ now
- Algorithm: Enter with $\underline{x}(k) =$ current state measurement
 $u(k-1) =$ last control generated
 - Need to set up a delay stack (initialized to zero)
$$V = [v_0, v_1, \dots, v_M] = [u(k-1-M) \quad u(k-M) \quad \dots \quad u(k-1)]$$
 - Propagate current state ahead ε sec: $\underline{x}_e = \hat{\underline{x}}(kh + \varepsilon)$

$$\hat{\underline{x}}(kh + \varepsilon) = e^{A\varepsilon} \underline{x}(k) + \int_0^\varepsilon e^{A\sigma} d\sigma B u(k-1-M)$$
 - Propagate \underline{x}_e ahead M time steps and apply control $u = -K\underline{x}_e$



- Algorithm can be rearranged for greater efficiency (need to store $\Phi \Gamma$)



Comparison with Smith Predictor Structure ($\epsilon=0$)

- Define system "model"

$$\underline{\xi}(k+1) = \Phi \underline{\xi}(k) + \Gamma u(k)$$

$\underline{\xi}(k) \sim$ crude estimate of $\underline{x}(k + M)$

$$\Rightarrow \underline{\xi}(k) = \Phi^M \underline{\xi}(k - M) + \sum_{i=1}^M \Phi^{i-1} \Gamma u(k - i)$$

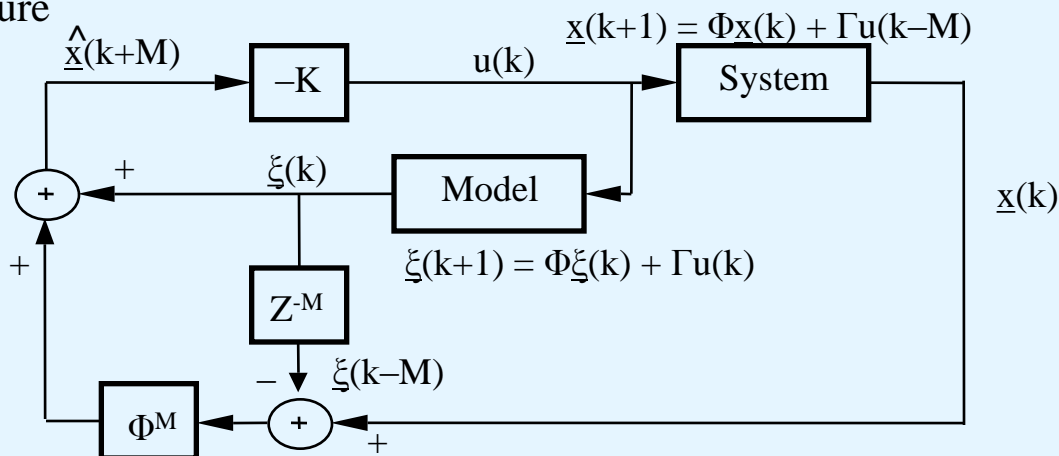
$$\Rightarrow \sum_{i=1}^M \Phi^{i-1} \Gamma u(k - i) = \underline{\xi}(k) - \Phi^M \underline{\xi}(k - M)$$

- $\hat{\underline{x}}(k+M)$ = Prediction estimate:

$$\hat{\underline{x}}(k+M) = \Phi^M [\underline{x}(k) - \underline{\xi}(k-M)] + \underline{\xi}(k)$$

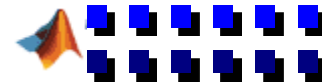
$$\text{control } u(k) = -K \hat{\underline{x}}(k + M)$$

- Loop structure



- Nearly identical to Smith predictor ($\underline{x} \sim y$)

- Preferable to use state propagation formula, especially if system is open-loop unstable and Φ, Γ are not perfectly known.



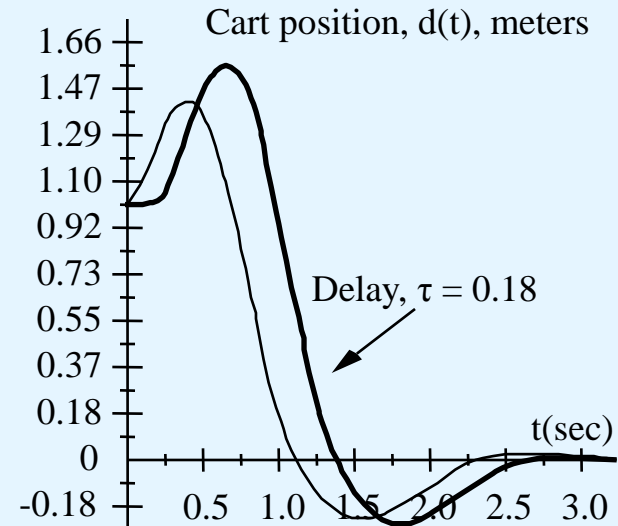
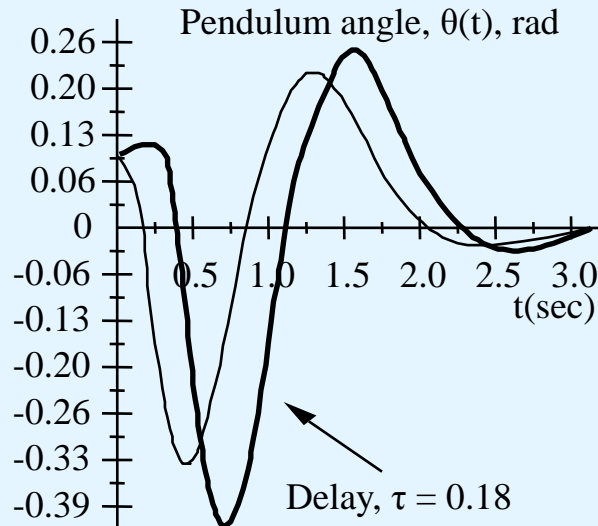


Example – Inverted Pendulum

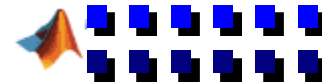
$h = 0.18$ sec $K = [-43.8 \quad -13.2 \quad -6.67 \quad -5.91]$ (gains obtained via pole placement)

$$\omega_c = 6.7, \phi_m = 19.3^\circ \Rightarrow \tau_{\max} = \phi_m / \omega_c \approx 0.05 \text{ sec}$$

- Select $\tau = 0.18$ (corresponds to $M = 1, \varepsilon = 0$).
System is highly unstable unless delay is compensated.
- Simulation $\underline{x}(0) = [0.1 \quad 0 \quad 1.0 \quad 0] = [\theta, \dot{\theta}, d, \dot{d}]$
compare with response of system with no delay



- System "drifts" for first τ sec, then is controlled to zero.
- In ideal case, state response for $k > M$ is identical to an undelayed response with an initial condition $\underline{x}(M) = \Phi^M \underline{x}(0)$, and shifted by Mh sec.
 \Rightarrow from $k \geq M$, predictor control is "perfect" (assuming you know Φ and Γ).





Robust MIMO Pole Placement : State Feedback - 1

- Kautsky's Algorithm
- In MIMO, we have mn degrees of freedom, but only n pole locations. Use the remaining degrees of freedom to *minimize the conditioning of the closed-loop eigen vector matrix*.

Recall Eigen value conditioning:

$$s_j = \frac{d\lambda_j}{d\varepsilon} = \frac{\| \underline{y}_j \|_2 \| \underline{x}_j \|_2}{| \underline{y}_j^T \underline{x}_j |}; \varepsilon = \text{parameter in } \Phi, \Gamma, K$$

$\underline{x}_j =$ Right eigen vector of $\Phi - \Gamma K$; $\underline{y}_j =$ Left eigen vector of $\Phi - \Gamma K$

$$\text{Metrics : (i) } J = \min_K \max_j s_j; \text{ (ii) } J = \min_K \left(\sum_{i=1}^n s_j^2 \right)^{1/2}; \text{ (iii) } \kappa_2(T) = \|T\|_2 \|T^{-1}\|_2$$

Let $\Lambda = \text{diag}(\lambda_1, \lambda_2, \dots, \lambda_n)$ be pole locations

If T is the matrix of eigen vectors: $(\Phi - \Gamma K)T = T\Lambda \Rightarrow \Gamma K = \Phi - T\Lambda T^{-1}$

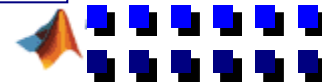
Suppose we do QR decomposition of $\Gamma : \Gamma = Q_1 R_1 = [Q_1 \quad Q_2] \begin{bmatrix} R_1 \\ 0 \end{bmatrix}$

so, $R_1 K = Q_1^T (\Phi - T\Lambda T^{-1}) \Rightarrow K = R_1^{-1} Q_1^T (\Phi - T\Lambda T^{-1})$. Also, $Q_2^T \Gamma K = 0 = Q_2^T (\Phi - T\Lambda T^{-1})$

One way of picking T is find $N \left[(\Phi - \lambda_j I)^T Q_2 \right]; j = 1, 2, \dots, n$ and pick directions from these.

So, $(\Phi - \lambda_j I)^T Q_2 = [L_j \quad V_j] \begin{bmatrix} \Delta_j \\ 0 \end{bmatrix} = L_j \Delta_j \Rightarrow V_j \in N \left[(\Phi - \lambda_j I)^T Q_2 \right]$

One approach: Form an arbitrary T . Replace column $t_j \ni$ new $t_j = \frac{V_j V_j^T \underline{u}_j}{\|V_j^T \underline{u}_j\|_2}$ where $\underline{u}_j \perp R[t_1, t_2, \dots, t_{j-1}, t_{j+1}, \dots, t_n]$





Robust Pole Placement Algorithm

- Kautsky's Algorithm

- Step 1: Do QR decomposition of $\Gamma = Q_1 R_1 = [Q_1 \quad Q_2] \begin{bmatrix} R_1 \\ 0 \end{bmatrix}$

- Step 2: For $j=1,2,\dots,n$ Do

Compute V_j by doing QR decomposition of

$$\left[(\Phi - \lambda_j I)^T Q_2 \right] = [L_j \quad V_j] \begin{bmatrix} \Delta_j \\ 0 \end{bmatrix}$$

End

- Step 3: Select from $\{V_j\}_{j=1}^n$ one from each an independent vector set to form T .

Now iterate until $k_2(T)$ stabilizes.

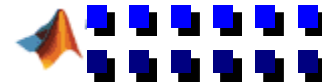
$$[\underline{t}_1, \underline{t}_2, \dots, \underline{t}_{j-1}, \underline{t}_{j+1}, \dots, \underline{t}_n] = [U_j \quad \underline{u}_j] \begin{bmatrix} Z_j \\ 0 \end{bmatrix} \Rightarrow new \underline{t}_j = \frac{V_j V_j^T \underline{u}_j}{\|V_j^T \underline{u}_j\|_2}$$

Note: $s_j = \frac{1}{|\underline{u}_j^T \underline{t}_j|} = \frac{1}{\|V_j^T \underline{u}_j\|_2}$

MATLAB place
implements this

- Compute gains:

$$K = R_1^{-1} Q_1^T (\Phi - T \Lambda T^{-1})$$



Application Example

- Chemical Reactor Example: $n=4$, $m=2$

- Continuous-time system

$$A = \begin{bmatrix} 1.3800 & -0.2077 & 6.7150 & -5.6760 \\ -0.5814 & -4.2900 & 0 & 0.6750 \\ 1.0670 & 4.2730 & -6.6540 & 5.8930 \\ 0.0480 & 4.2730 & 1.3430 & -2.1040 \end{bmatrix}; B = \begin{bmatrix} 0 & 5.6790 \\ 1.1360 & 1.1360 \\ 0 & 0 \\ -3.1460 & 0 \end{bmatrix}; \lambda_i(A) = \begin{bmatrix} 1.9910 \\ 0.0635 \\ -5.0566 \\ -8.6659 \end{bmatrix}; \text{desired } \lambda_i(\bar{A}) = \begin{bmatrix} -0.200 \\ -0.500 \\ -5.0566 \\ -8.6659 \end{bmatrix}$$

- $h = 0.2 / \|A\| = 0.0154 \Rightarrow$ select $h = 0.01$ sec

- Discretized system

$$\Phi = \begin{bmatrix} 1.0142 & -0.0018 & 0.0651 & -0.0546 \\ -0.0057 & 0.9582 & -0.0001 & 0.0067 \\ 0.0103 & 0.0417 & 0.9363 & 0.0563 \\ 0.0004 & 0.0417 & 0.0129 & 0.9797 \end{bmatrix}; \Gamma = \begin{bmatrix} 0.0009 & 0.0572 \\ 0.0110 & 0.0110 \\ -0.0007 & 0.0005 \\ -0.0309 & 0.0003 \end{bmatrix}; \lambda_i(\Phi) = \begin{bmatrix} 1.0201 \\ 1.0006 \\ 0.9507 \\ 0.9170 \end{bmatrix}; \text{desired } \lambda_i(\bar{\Phi}) = \begin{bmatrix} 0.9980 \\ 0.9950 \\ 0.9507 \\ 0.9170 \end{bmatrix}$$

- Gains $K = \begin{bmatrix} 0.1325 & -0.9186 & 0.1731 & -0.0692 \\ 0.5030 & 0.6180 & 0.4966 & -0.3457 \end{bmatrix}; T = \begin{bmatrix} -0.5261 & 0.0834 & 0.7273 & -0.5877 \\ 0.0428 & 0.8082 & -0.2123 & 0.2540 \\ 0.8214 & 0.2057 & 0.4402 & 0.5555 \\ -0.2162 & -0.5455 & 0.4819 & 0.5306 \end{bmatrix}; \kappa_2(T) = 2.5237$



Summary of Pole Placement Design by SVFB

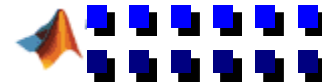
- Valid for continuous or discrete design
 - Ackermann formula to find K
 - Transform $K \rightarrow \tilde{K}$ if design developed on C and $h \sim$ small
 - Need to select all n pole locations
- SVFB does not modify system zeros
 - Can combine compensator $H(z)$ and SVFB to adjust/move zeros

=> Advantages

- Straightforward design methodology
- Direct control over CL pole locations
- Uses all available information in the feedback
- Ability to design deadbeat control
- Possible to extend to MIMO systems, but cumbersome

=> Disadvantages

- Need to measure or estimate all states
- More complex design than series compensation
- No direct control over CL time response (still requires trial and error with CL simulation)
- Not always clear where to place all n poles
- No direct control over ϕ_m, ω_c





Command Inputs to SVFB Systems

- Consider continuous case for simplicity

$$\dot{\underline{x}}(t) = A \underline{x}(t) + B u(t)$$

$$u(t) = -K \underline{x}(t)$$

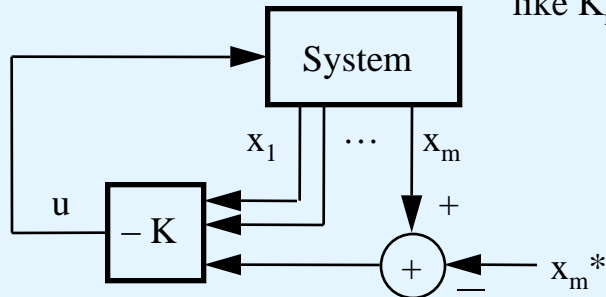
- Closed-loop $A - BK$ has poles at desired locations

- Desire $x_m = m$ -th component of $\underline{x}(t) \rightarrow x_m^*$ in steady-state

- Idea: Use control $u(t) = -K_1 x_1(t) - \dots - K_m [x_m(t) - x_m^*] - \dots - K_n x_n(t)$

$$u(t) = \underbrace{K_m x_m^*}_{\text{like } K_1 r(t) \text{ with } r(t) = \text{constant}} - K \underline{x}(t)$$

like $K_1 r(t)$ with $r(t) = \text{constant}$



$$\dot{\underline{x}}(t) = (A - BK) \underline{x}(t) + BK_m x_m^*$$

- Steady-state \underline{x} : $\underline{x}_{ss} = -(A - BK)^{-1} B K_m x_m^*$

$$\underline{x}_{m,ss} = -\underline{e}_m' (A - BK)^{-1} B K_m x_m^*; \quad \underline{e}_m' = [0 \ 0 \ \dots \ 1 \ \dots \ 0]$$

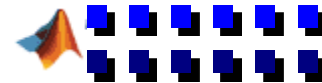
m-th element

- Generally $x_{m,ss} \neq x_m^*$

=> Adjust $x_m^* \rightarrow \beta x_m^*$ and pick β so that $x_{m,ss} = x_m^*$

or consider use of integral control

$$u(t) = -K x_1(t) - \dots - K_m [x_m(t) - x_m^*] - \dots - K_n x_n(t) - K_{n+1} \int_0^t [x_m(\sigma) - x_m^*] d\sigma$$





Integral Control in SVFB

- Define $x_m(t)|_{\text{new}} = x_m(t)|_{\text{old}} - x_m^* \triangleq$ error in state $x_m \Rightarrow \dot{\underline{x}}(t) = A \underline{x}(t) + \underline{a}_m x_m^* + B u(t)$
 $\underline{a}_m = m$ -th column of A , (often $\underline{a}_m = \underline{0}$, especially if x_m is a position variable)

- Define $x_{n+1}(t) = \int_0^t x_m(\sigma) d\sigma = \begin{cases} \text{integral of error in } x_m \\ \text{from desired ss value} \end{cases}$
 $\dot{x}_{n+1}(t) = x_m(t); \quad x_{n+1}(0) = 0$

- Augmented $(n+1)$ -st order system, $\underline{x}_a = [\underline{x}, x_{n+1}]'$

$$\begin{bmatrix} \dot{\underline{x}} \\ \dot{x}_{n+1} \end{bmatrix} = \begin{bmatrix} A & 0 \\ \underline{e}_m' & 0 \end{bmatrix} \begin{bmatrix} \underline{x} \\ x_{n+1} \end{bmatrix} + \begin{bmatrix} \underline{a}_m \\ 0 \end{bmatrix} x_m^* + \begin{bmatrix} B \\ 0 \end{bmatrix} u$$

$$\dot{\underline{x}}_a(t) = A_a \underline{x}_a(t) + B_a u(t) + \tilde{\underline{a}}_m x_m^*$$

- Augmented system may not be controllable. Examine

$$[B_a \quad A_a B_a \quad \cdots \quad A_a^n B_a] = \begin{bmatrix} B & AB & \cdots & A^n B \\ 0 & \underline{e}_m' B & \cdots & \underline{e}_m' A^{n-1} B \end{bmatrix} = (n+1) \times (n+1) \quad \text{Controllability matrix}$$

- Selection of $u(t) = -K_a \underline{x}_a(t) = -K \underline{x}(t) - K_{n+1} x_{n+1}(t)$

- Design K as before, to place poles of $A - BK = \bar{A}$

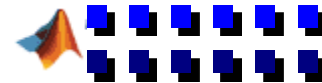
- $K_{n+1} \sim$ small gain on integral error

- CL characteristic polynomial, $|sI - A_a + B_a K_a|$

$$\begin{vmatrix} sI - \bar{A} & BK_{n+1} \\ -\underline{e}_m' & s \end{vmatrix} = |sI - \bar{A}| \cdot \underbrace{|s + \underline{e}_m' (sI - \bar{A})^{-1} B K_{n+1}|}_{\sim (s - \underline{e}_m' \bar{A}^{-1} B K_{n+1}) \text{ for small } K_{n+1}} = |sI - \bar{A}|$$

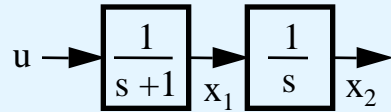
$$\sim (s - \underline{e}_m' \bar{A}^{-1} B K_{n+1}) \text{ for small } K_{n+1}$$

- Select K_{n+1} so that pole is in LHP.





Example – Integral Control



x_1 = Motor shaft velocity
 x_2 = Shaft angular position

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -1 & 0 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u$$

$u = -[1 \ 2] \underline{x}$ places CL poles at $s = -1 \pm j$

(1) Desire $x_1(t) \rightarrow x_1^*$ in ss (obvious problem since $x_2 \rightarrow \infty$)

Introduce integral control, $x_1 \triangleq x_{1,e}$, obtain augmented state equation

$$\dot{\underline{x}}_a = \begin{bmatrix} -1 & 0 & 0 \\ 1 & 0 & 0 \\ 1 & 0 & 0 \end{bmatrix} \underline{x}_a + \begin{bmatrix} -1 \\ 1 \\ 0 \end{bmatrix} x_1^* + \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} u ; |H_c| = \begin{vmatrix} 1 & -1 & 1 \\ 0 & 1 & -1 \\ 0 & 1 & -1 \end{vmatrix} = 0!$$

- Uncontrollable, cannot have a stable system if $x_1 = \text{constant}$

(2) Desire $x_2(t) \rightarrow x_2^*$ in ss. Introduce integral control, $x_2 \rightarrow x_{2,e}$

$$\dot{\underline{x}}_a = \begin{bmatrix} -1 & 0 & 0 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \underline{x}_a + \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} x_2^* + \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} u ; |H_c| \neq 0$$

$$u = -[K_1 \ K_2 \ K_3] \underline{x}_a = -K_1 x_1 - K_2 x_2 - K_3 \int_0^t x_2(\sigma) d\sigma$$

let $[K_1 \ K_2] = [1 \ 2]$ = same as before for primary poles; $K_3 = \epsilon \sim \text{small}$

$$|sI - \bar{A}_a| = s(s^2 + 2s + 2) + \epsilon \sim (s^2 + 2s + 2)(s + \epsilon/2)$$

- Too much gain on \int -term is NG when simply added in

Alternate approach - Pick $[K_1 \ K_2 \ K_3]$ to place all 3 CL poles in LHP -- but this destroys association $K_3 \leftrightarrow \text{new pole @ } s \sim 0$

